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Aerodromes Annex 14 to the Convention on International Civil Aviation International Civil Aviation Organization International Standards and Recommended Practices Third Edition July 2009 Volume II Heliports This editions of Annex 14, Volume I. For information regarding the applicability of Standards and Recommended Practices, Foreword and the relevant clauses in each chapter.see 5 March 2009 19 9 I 2. International Standards and Recommended Practices This edition incorporates all amendments adopted by the Council prior to 5 March 2009 and supersedes, on 19 November 2009, all previous editions of Annex 14, Volume II. For information regarding the applicability of the Standards and Recommended Practices, see Foreword and the relevant clauses in each chapter. Third Edition July 2009 International Civil Aviation Organization Aerodromes Annex 14 to the Convention on International Civil Aviation Volume II Heliports 3. (ii) Published in separate English, Arabic, Chinese, French, Russian and Spanish editions by the INTERNATIONAL CIVIL AVIATION ORGANIZATION 999 University Street, Montréal, Quebec, Canada H3C 5H7 For ordering information and for a complete listing of sales agents and booksellers, please go to the ICAO website at www.icao.int First edition 1990 Second edition 1995 Third edition 2009 Annex 14, Volume II, Heliports Order Number: AN14-2 ISBN 978-92-9231-330-2 © ICAO 2009 All rights reserved. No part of this publication may be reproduced, stored in a retrieval system or transmitted in any form or by any means, without prior permission in writing from the International Civil Aviation Organization. 4. (iii) AMENDMENTS Amendments are announced in the supplements to the Catalogue of ICAO Publications; the Catalogue and its supplements are available on the ICAO website at www.icao.int. The space below is provided to keep a record of such amendments. RECORD OF AMENDMENTS AND CORRIGENDA AMENDMENTS CORRIGENDA No. Date applicable Date entered Entered by No. Date of issue Date entered Entered by 1-4 Incorporated in this edition 5. ANNEX 14 - VOLUME II (v) 19/11/09 TABLE OF CONTENTS Page Abbreviations and symbols; manuals (viii) FOREWORD. (ix) CHAPTER 1. General.... . 1-1 1.1 Definitions... .. 1-1 1.2 Applicability .. 1-4 1.3.1 Horizontal reference system 1-4 1.3.2 Vertical reference 1-3 1.3 Common reference systems 1-4 1.3.3 Temporal reference system.... 1-4 CHAPTER 2. Heliport data.. 2-1 2.1 Aeronautical data system.. 2-2 2.3 Heliport elevation..... ... 2-2 2.4 Heliport dimensions and related . 2-1 2.2 Heliport reference point 2-2 2.5 Declared distances. . 2-3 2.6 Coordination between aeronautical information services and heliport authorities..... . 2-4 CHAPTER 3. 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Foot HAPI Helicopter approach path indicator Hz Hertz IMC Instrument meteorological conditions kg Kilogram km/h Kilometre per hour kt Knot L Litre LDAH Landing distance available L/min Litre per minute m Metre RD Diameter of the largest rotor RTODAH Rejected take-off distance available s Second TLOF Touchdown and lift-off area TODAH Take-off distance available VMC Visual meteorological conditions Symbols ° Degree = Equals % Percentage ± Plus or minus MANUALS (related to the specifications of this Annex) Aerodrome Design Manual (Doc 9157) Part 1 — Runways Part 2 — Taxiways, Aprons and Holding Bays Part 3 — Pavements Part 4 — Visual Aids Part 5 — Electrical Systems Part 6 — Frangibility Airport Planning Manual (Doc 9184) Part 1 — Master Planning Part 2 — Land Use and Environmental Control Part 3 — Guidelines for Consultant/Construction Services Manual (Doc 9137) Part 1 — Rescue and Fire Fighting Part 2 — Pavement Surface Conditions Part 3 — Bird Control and Reduction Part 4 — Fog Dispersal (withdrawn) Part 5 — Removal of Disabled Aircraft Part 6 — Control of Obstacles Part 7 — Airport Emergency Planning Part 8 — Airport Maintenance Practices Heliport Manual (Doc 9261) Manual of Surface Movement Guidance and Control Systems (SMGCS) (Doc 9476) Manual on the ICAO Bird Strike Information System (IBIS) (Doc 9332) Stolport Manual (Doc 9150) 9. ANNEX 14 - VOLUME II (ix) 19/11/09 FOREWORD Historical background Standards and Recommended Practices for aerodromes were first adopted by the Council on 29 May 1951 pursuant to the provisions of Article 37 of the Convention on International Civil Aviation (Chicago 1944) and designated as Annex 14 to the Convention. In general, Volume I addresses planning, design and operations of aerodromes but is not specifically applicable to heliports. Therefore, Volume II is being introduced as a means of including provisions for heliports. Proposals for comprehensive Standards and Recommended Practices covering all aspects of heliport planning, design and operations have been developed with the assistance of the ANC Visual Aids Panel and the ANC Helicopter Operations Panel. Table A shows the origin of the provisions in this volume, together with a list of the principal subjects involved and the dates on which the Annex was adopted by the Council, when it became applicable. Action by Contracting States Notification of differences. The attention of Contracting States is drawn to the obligation imposed by Article 38 of the Convention by which Contracting States are required to notify the Organization of any differences between their national regulations and practices and the International States are invited to extend such notification to any differences from Recommended Practices contained in this Annex and any amendments thereto, when the notification of such differences is important for the safety of air navigation. Further, Contracting States are invited to keep the Organization currently informed of any differences which may subsequently occur, or of the withdrawal of any differences previously notified. A specified request for notification of differences will be sent to Contracting States immediately after the adoption of each amendment to this Annex. The attention of States is also drawn to the publication of differences between their national regulations and practices and the related ICAO Standards and Recommended Practices through the Aeronautical Information Service, in addition to the obligation of States under Article 38 of the Convention. Promulgation of information. The establishment and withdrawal of and changes to facilities, services and procedures affecting aircraft operations provided in accordance with the Standards and Recommended Practices specified in this Annex should be notified and take effect in accordance with the provisions of Annex 15. Status of Annex components An Annex is made up of the following component parts, not all of which, however, are necessarily found in every Annex; they have the status indicated: 1. — Material comprising the Annex proper: a) Standards and Recommended Practices adopted by the Council under the provisions of the Convention. They are defined as follows: 10. Annex 14 — Aerodromes Volume II 19/11/09 (x) Standard: Any specification for physical characteristics, configuration, matériel, performance, personnel or procedure, the uniform application of which is recognized as necessary for the safety or regularity of international air navigation and to which Contracting States will conform in accordance with the Convention; in the event of impossibility of compliance, notification for physical characteristics, configuration, material, performance, personnel or procedure, the uniform application of which is recognized as desirable in the interest of safety, regularity or efficiency of international air navigation, and to which Contracting States will endeavour to conform in accordance with the Convention. b) Appendices comprising material grouped separately for convenience but forming part of the Standards and Recommended Practices adopted by the Council. c) Definitions of terms used in the Standards and Recommended Practice in which the term is used, since a change in the meaning of the term would affect the specifications. d) Tables and Figures which add to or illustrate a Standard or Recommended Practice and have the same status. 2.— Material approved by the Council for publication in association with the Standards and Recommended Practices: a) Forewords comprising historical and explanatory material based on the action of the obligations of States with regard to the application of the Standards and Recommended Practices ensuing from the Convention and the Resolution of Adoption. b) Introductions comprising explanatory material introduced at the beginning of parts, chapters or sections of the Annex to assist in the understanding of the application of the text. c) Notes included in the text, where appropriate, to give factual information or references bearing on the Standards or Recommended Practices in question, but not constituting part of the Standards or Recommended Practices, d) Attachments comprising material supplementary to the Standards and Recommended Practices, or included as a guide to their application. Selection of languages — English, Arabic, Chinese, French, Russian and Spanish. Each Contracting State is requested to select one of those texts for the purpose of national implementation and for other effects provided for in the Convention, either through direct use or through translation into its own national language, and to notify the Organization accordingly. Editorial practices The following practice has been adhered to in order to indicate at a glance the status of each statement: Standards have been printed in light face italics, the status being indicated by the prefix Note. 11. Foreword Annex 14 — Aerodromes (xi) 19/11/09 The following editorial practice has been followed in the writing of specifications: for Standards the operative verb "should" is used. The units of measurement used in this document are in accordance with the International System of Units (SI) as specified in Annex 5 to the Convention on International Civil Aviation. Where Annex 5 permits the use of non-SI alternative units these are shown in parentheses following the basic units. Where two sets of units are quoted it must not be assumed that the pairs of values are equal and interchangeable. It may, however, be inferred that an equivalent level of safety is achieved when either set of units is used exclusively. Any reference to a portion of this document, which is identified by a number and/or title, includes all subdivisions of that portion. Table A. Amendments to Annex 14. Volume II Amendments to Annex 14. Volume II Amendment Source(s) Subject(s) Adopted Effective Applicable 1st Edition Fourth Meeting of the ANC Helicopter Operations Panel: Eleventh Meeting of the ANC Visual Aids Panel and Secretariat Physical characteristics; obstacle limitation surfaces; visual aids for visual meteorological conditions; rescue and fire fighting services. 19 March 1990 15 November 1990 1 (2nd Edition) Twelfth Meeting of the ANC Visual Aids Panel and Secretariat Standard geodetic reference system (WGS-84); frangibility; visual aids for helicopter non-precision approaches; and visual alignment guidance system. 13 March 1995 24 July 1995 19 November 1995 24 July 1995 19 November 1995 2 Air Navigation Commission Aeronautical databases and vertical component of the World Geodetic System — 1984 (WGS-84). 21 March 1997 21 July 1997 16 November 1997 3 Fourteenth Meeting of the ANC Visual Aids Panel and Secretariat Definitions of calendar, datum, Gregorian calendar and obstacle; common reference systems; heliport dimensions and related information; touchdown and lift-off area lighting system; Appendix 1 — Aeronautical Data Quality Requirements. 27 February 2004 12 July 2004 25 November 2004 4 (3rd Edition) First Meeting of the Aerodromes Panel Introductory note; definitions of air transit route, declared distances, dynamic load-bearing surface, final approach and take-off area, helicopter air taxiway, helicopter stand, helideck, obstacle, protection area, rejected take-off area shipboard heliports, static load-bearing surface, taxi-route, touchdown and lift-off area, winching area; applicability; physical characteristics for surface-level heliports, helidecks, and shipboard heliports; obstacle limitation surfaces and sectors and requirements for helidecks and shipboard heliports; winching area marking; heliport identification marking; maximum allowable mass marking; helideck obstacle-free sector marking; touchdown and lift-off area marking; helideck surface marking; helideck surfac 12. ANNEX 14 — VOLUME II 1-1 19/11/09 INTERNATIONAL STANDARDS AND RECOMMENDED PRACTICES CHAPTER 1. GENERAL Introductory Note.— Annex 14, Volume II, contains Standards and Recommended Practices (specifications) that prescribe the physical characteristics and obstacle limitation surfaces to be provided for at heliports, and certain facilities and technical services normally provided at a heliport. It is not intended that these specifications limit or regulate the operation of an aircraft. When designing a heliport, the critical design helicopter, having the largest set of dimensions and the greatest maximum take-off mass (MTOM) the heliport is intended to serve, would need to be considered. It is to be noted that provisions for helicopter flight operations are contained in Annex 6, Part III. 1.1 Definitions When the following terms are used in both volumes. Accuracy. A degree of conformance neasured value and the true value. Note.— For measured positional data, the accuracy is normally expressed in terms of a distance from a stated position within which there is a defined confidence of the true position falling. Air transit route. A defined route for the air transiting of helicopters. Calendar. Discrete tempora reference system that provides the basis for defining temporal position to a resolution of data that provides a level of assurance against loss or alteration of data. Data quality. A degree or level of confidence that the data provided meet the requirements of the data user in terms of accuracy, resolution and integrity. Datum. Any quantities that may serve as a reference or basis for the calculation of other quantities (ISO 19104**). Declared distances — heliports. a) Take-off distance available (TODAH). The length of the FATO plus the length of helicopter clearway (if provided) declared available and suitable for helicopters to complete the take-off. * ISO Standard 19104, Geographic information — Terminology 13. Annex 14 — Aerodromes Volume II 19/11/09 1-2 b) Rejected take-off distance available (RTODAH). The length of the FATO declared available and suitable for helicopters operated in performance class 1 to complete a rejected take-off. c) Landing distance available (LDAH). The length of the FATO plus any additional area declared available for helicopters to complete the landing manoeuvre from a defined height. Dynamic load-bearing surface. A surface capable of supporting the loads generated by a helicopter conducting an emergency touchdown on it. Elevated heliport. A heliport located on a raised structure on land. Ellipsoid, measured along the ellipsoid, measured along the ellipsoid height (Geodetic height). The height related to the reference ellipsoid, measured along the ellipsoid height (Geodetic height). (FATO). A defined area over which the final phase of the approach manoeuvre to hover or landing is completed and from which the take-off manoeuvre is commenced. Where the FATO is to be used by helicopters operated in performance class 1, the defined area includes the rejected take-off manoeuvre is commenced. Where the FATO is to be used by helicopters operated in performance class 1, the defined area includes the rejected take-off manoeuvre is commenced. parameters required to define location and orientation of the local reference system with respect to the global reference system/frame. Geoid. The equipotential surface in the gravity field of the Earth which coincides with the undisturbed mean sea level (MSL) extended continuously through the continents. Note.— The geoid is irregular in shape because of local gravitational disturbances (wind tides, salinity, current, etc.) and the direction of gravity is perpendicular to the geoid at every point. Geoid undulation. The distance of the geoid at every point. Geoid undulation. The distance of the geoid at every point. ellipsoid, the difference between the WGS-84 ellipsoidal height and orthometric height represents WGS-84 geoid undulation. Gregorian calendar, common years that more closely approximates the tropical year that the tropical year that have 365 days and leap years 366 days divided into twelve sequential months. Helicopter air taxiway. A defined area on the ground or water, selected and/or prepared as a suitable area over which a helicopter operated in performance class 1 may accelerate and achieve a specific height. Helicopter stand which provides for parking a helicopter and where ground taxiway intended for the ground taxiway intended for taxiwa operations. Helideck. A heliport located on an offshore structure such as an exploration or production platform used for the exploitation of oil or gas. *** ISO Standard 19108, Geographic information — Temporal schema 14. Chapter 1 Annex 14 — Aerodromes 1-3 19/11/09 Heliport. An aerodrome or a defined area on a structure intended to be used wholly or in part for the arrival, departure and surface movement of helicopters. Integrity (aeronautical data and its value has not been lost nor authorized amendment. Obstacle. All fixed (whether temporary or permanent) and mobile objects, or parts thereof, that: a) are located on an area intended for the surface movement of aircraft; or b) extend above a defined surface intended to protect aircraft in flight; or c) stand outside those defined surfaces and that have been assessed as being a hazard to air navigation. Orthometric height of a point related to the geoid, generally presented as an MSL elevation. Protection area. An area within a taxi-route and helicopter stands, for safe manoeuvring of helicopters stands, for safe manoeuvring of helicopters and helicopters and helicopters. Rejected take-off area. A defined area on a heliport suitable for helicopters. Rejected take-off area. off. Safety area. A defined area on a heliport surrounding the FATO which is free of obstacles, other than those required for air navigation purpose-built. A purpose-built diverging from the FATO. shipboard heliport is one designed specifically for helicopter but not designed specifically for that task. Static load-bearing surface. A surface capable of supporting the mass of a helicopter situated upon it. Station declination. An alignment variation between the zero degree radial of a VOR and true north, determined at the time the VOR station is calibrated. Surface-level heliport. A heliport to another. A taxi-route includes a helicopter air or ground taxiway which is centred on the transfer by helicopter of personnel or stores to or from a ship. 1.2 Applicability Note.— The dimensions discussed in this Annex are based on consideration of single-main-rotor helicopters. For tandem-rotor helicopters the heliport design will be based on a case-by-case review of the specific models using the basic requirement for a safety area and protection areas specified in this Annex. 1.2.1 The interpretation of some of the specifications in the Annex expressly requires the exercising of discretion, the taking of a decision or the performance of a function by the appropriate authority. In other specifications, the expression 15. Annex 14 — Aerodromes Volume II 19/11/09 1-4 appropriate authority does not actually appear although its inclusion is implied. shall rest with the State having jurisdiction over the heliport. 1.2.2 The specifications in Annex 14, Volume II, shall apply to all heliports international civil aviation. They shall apply equally to areas for the exclusive use of helicopters at an aerodrome primarily meant for the use of aeroplanes. Where relevant, the provisions of Annex 14, Volume I, shall apply to the helicopter operations being conducted at such an aerodrome. 1.2.3 Unless otherwise specified, the specification for a colour reference systems 1.3.1.1 World Geodetic System — 1984 (WGS-84) shall be used as the horizontal (geodetic) reference datum. Note.— Comprehensive guidance material concerning WGS-84 is contained in the World Geodetic System — 1984 (WGS-84) Manual (Doc 9674). 1.3.2 Vertical reference system 1.3.2.1 Mean sea level (MSL) datum, which gives the relationship of gravity-related height (elevation) to a surface known as the geoid, shall be used as the vertical reference system. Note 1.— The geoid globally most closely approximates MSL. It is defined as the equipotential surface in the gravity field of the Earth which coincides with the undisturbed MSL extended continuously through the continents. Note 2.— Gravity-related heights. 1.3.3 Temporal reference system 1.3.3.1 The Gregorian calendar and Coordinated Universal Time (UTC) shall be used as the temporal reference system. 1.3.3.2 When a different temporal reference system is used, this shall be indicated in GEN 2.1.2 of the Aeronautical Information Publication (AIP). 16. ANNEX 14 — VOLUME II 2-1 19/11/09 CHAPTER 2. HELIPORT DATA 2.1 Aeronautical data 2.1.1 Determination and reporting of heliport-related aeronautical data shall be in accordance with the accuracy requirements for aeronautical data are based upon a 95 per cent confidence level and in that respect, three types of positional data shall be identified: surveyed points (e.g. FATO threshold), calculated points (e.g. flight information region boundary points). Note. — Specifications governing the quality system are given in Annex 15, Chapter 3. 2.1.2 Contracting States shall ensure that integrity of aeronautical data is maintained throughout the data process from survey/origin to the next intended user. Aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data is maintained throughout the data process from survey/origin to the next integrity of aeronautical data process from survey/origin to the n the use to which the data item is put. Consequently, the following classifications and data integrity levels shall apply: a) critical data, integrity level 1 × 10-8 : there is a high probability when using corrupted critical data, integrity level 1 × 10-8 : there is a high probability when using corrupted critical data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe; b) essential data, integrity level 1 × 10-5 : there is a low probability when using corrupted essential data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe; and c) routine data, integrity level 1 × 10-3 : there is a very low probability when using corrupted routine data that the continued safe flight and landing of an aircraft would be severely at risk with the potential for catastrophe. 2.1.3 Protection of electronic aeronautical data while stored or in transit shall be totally monitored by the cyclic redundancy check (CRC). To achieve protection of the integrity level of critical and essential aeronautical data while stored or in transit shall be totally monitored by the cyclic redundancy check (CRC). algorithm shall apply respectively. 2.1.4 Recommendation.— To achieve protection of the integrity level of routine aeronautical data as classified in 2.1.2, a 16-bit CRC algorithm should apply. Note.— Guidance material on the aeronautical data quality requirements (accuracy, resolution, integrity, protection and traceability) is contained in the World Geodetic System — 1984 (WGS-84) Manual (Doc 9674). Supporting material in respect of the provisions of Appendix 1 related to accuracy and integrity of aeronautical data is contained in RTCA Document ED-77 — Industry Requirements for Aeronautical Information. 2.1.5 Geographical coordinates indicating latitude and longitude shall be determined and reported to the aeronautical information services authority in terms of the World Geodetic System - 1984 (WGS-84) geodetic reference datum, identifying those geographical coordinates which have been transformed into WGS-84 coordinates by mathematical means and whose accuracy of original field work does not meet the requirements in Appendix 1, Table A1-1. 2.1.6 The order of accuracy of the field work shall be such that the resulting operational navigation data for the phases of flight will be within the maximum deviations, with respect to an appropriate reference frame, as indicated in the tables contained in Appendix 1. 17. Annex 14 — Aerodromes Volume II 19/11/09 2-2 2.1.7 In addition to the elevation (referenced to the WGS-84 ellipsoid) for those positions as indicated in Appendix 1 shall be determined and reported to the aeronautical information services authority. Note 1.— An appropriate reference frame is that which enables WGS-84 to be realized on a given heliport and with respect to which all coordinates are given in Annex 15, Chapter 3. 2.2 Heliport reference point 2.2.1 A heliport reference point shall be established for a heliport not collocated with an aerodrome. Note.— When the heliport is collocated with an aerodrome reference point shall be located near the initial or planned geometric centre of the heliport and shall normally remain where first established. 2.2.3 The position of the heliport elevation and geoid undulation at the heliport elevation at the heliport elevation and geoid undulation at the heliport elevation a position shall be measured and reported to the aeronautical information services authority to the accuracy of one-half metre or foot. 2.3.2 For a heliport used by international civil aviation, the elevation of the TLOF and/or the elevation of the TLOF and/or the elevation and geoid undulation of the TLOF and/or the elevation of the TLOF and/or the elevation and geoid undulation of the TLOF and/or the elevation and geoid undulation of the TLOF and/or the elevation and geoid undulation of the TLOF and/or the elevation and geoid undulation of the TLOF and/or the elevation and geoid undulation of the TLOF and/or the elevation and geoid undulation of the TLOF and/or the elevation and geoid undulation of the the elevation and geoid undulation and geoid undulati aeronautical information services authority to the accuracy of: a) one-half metre or foot for non-precision approaches; and b) one-quarter metre or foot for precision approaches. Note.— Geoid undulation must be measured in accordance with the appropriate system of coordinates. 2.4 Heliport dimensions and related information 2.4.1 The following data shall be measured or described, as appropriate, for each facility provided on a heliport: a) heliport type — surface type, bearing strength in tonnes (1 000 kg); c) final approach and take-off area — type of FATO, true bearing to one-hundredth of a degree, designation number (where appropriate), length, width to the nearest metre or foot, slope, surface type; 18. Chapter 2 Annex 14 — Aerodromes 2-3 19/11/09 d) safety area — length, width and surface type; e) helicopter ground taxiway, air taxiway and air transit route — designation, width, surface type; f) apron — surface type, helicopter stands; g) clearway — length, ground profile; h) visual aids for approach procedures, marking and lighting of FATO, TLOF, taxiways and aprons; and i) distances to the nearest metre or foot of localizer and glide path elements comprising an instrument landing system (ILS) or azimuth and elevation antenna of a microwave landing system (MLS) in relation to the associated TLOF or FATO extremities. 2.4.2 The geographical coordinates of the geometric centre of the TLOF and/or of each threshold of the FATO (where appropriate) shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and hundredths of seconds. 2.4.3 The geographical coordinates of appropriate centre line points of helicopter ground taxiways, air taxiways and air transit routes shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and hundredths of seconds. aeronautical information services authority in degrees, minutes, seconds and tenths of seconds. 2.4.5 The geographical coordinates of obstacles in Area 3 shall be measured and reported to the aeronautical information services authority in degrees, minutes, seconds and tenths of seconds. In addition, the top elevation, type, marking and lighting (if any) of obstacles shall be reported to the aeronautical information services authority. Note 1.— See Annex 15, Appendix 8, for graphical illustrations of obstacle data collection surfaces and criteria used to identify obstacles in Areas 2 and 3. Note 2.— Appendix 1 to this Annex provides requirements for obstacle data determination in Areas 2 and 3. Note 3.— Implementation of Annex 15, provision 10.6.1.2, concerning the availability, as of 18 November 2010, of obstacle data according to Area 3 specifications would be facilitated by appropriate advance planning for the collection and processing of such data. 2.5 Declared distances The following distance available; and c) landing distance available; and c) landing distance available; b) rejected take-off distance available; b) rejected take-off distance available; and c) landing distance available; b) rejected take-off distance available; b) rejected 2.6.1 To ensure that aeronautical information services units obtain information to enable them to provide up-to-date pre-flight information, arrangements shall be made between aeronautical information, arrangements and heliport to the responsible for heliport to the responsible for heliport authorities responsible for heliport to the responsible for heliport authorities responsible for heliport to the responsible for heliport authorities responsible for heliport to the responsible for heliport authorities responsible for heliport authorities responsible for heliport authorities responsible for heliport to the responsible for heliport authorities responsible for helipor aeronautical information services unit, with a minimum of delay: a) information on heliport conditions; b) the operational status of associated facilities, services and navigation aids within their area of responsibility; c) any other information considered to be of operational significance. account shall be taken by the services responsible for such changes of the time needed by the aeronautical information service for the preparation, production and issue of relevant material for promulgation. To ensure timely provision of the information service for the preparation service for the preparation service for the preparation. therefore required. 2.6.3 Of a particular importance are changes to aeronautical information that affect charts and/or computer-based navigation system, as specified in Annex 15, Chapter 6 and Appendix 4. The predetermined, internationally agreed AIRAC effective dates in addition to 14 days postage time shall be observed by the responsible for the provision of raw aeronautical information/data to the aeronautical information/data to the aeronautical information/data to the aeronautical information services shall do that while taking into account accuracy and integrity requirements for aeronautical data as specified in Appendix 1 to this Annex. Note 1.— Specifications for the issue of a NOTAM and SNOWTAM are contained in Annex 15, Chapter 5, and Appendices 6 and 2, respectively. Note 2.— The AIRAC information is distributed by the AIS at least 42 days in advance of the AIRAC effective dates with the objective of reaching recipients at least 28 days in advance of the effective date. Note 3.— The schedule of the predetermined internationally agreed AIRAC common effective dates at intervals of 28 days, including 19 November 2009, and guidance for the AIRAC use are contained in the Aeronautical Information Services Manual (Doc 8126, Chapter 2, 2.6). 20. ANNEX 14 — VOLUME II 3-1 19/11/09 CHAPTER 3. PHYSICAL CHARACTERISTICS 3.1 Surface-level heliports only. Where a water heliport is being considered, the appropriate authority may establish suitable criteria. Note 2.— The dimensions of the taxi-routes and helicopter stands include a protection area. Final approach and take-off area (FATO). Note.— A FATO may be located on or near a runway strip or taxiway strip. 3.1.2 A FATO shall be obstacle free 3.1.3 The dimensions of a FATO shall be: a) where intended to be used by helicopters operated in performance class 1, as prescribed in the helicopter flight manual (HFM) except that, in the absence of width specifications, the width shall be not less than the greatest overall dimension (D) of the largest helicopter the FATO is intended to serve; b) where intended to be used by helicopters operated in performance class 2 or 3, of sufficient size and shape to contain an area within which can be drawn a circle of diameter not less than: 1) 1 D of the largest helicopters the FATO is intended to serve is more than 3 175 kg; 2) 0.83 D of the largest helicopter when the MTOM of helicopters the FATO is intended to serve is 3 175 kg or less. Note.— Where the term FATO is not used in the HFM, the minimum landing/take-off area specified in the HFM for the appropriate flight profile is used. 3.1.4 Recommendation.— Where intended to be used by helicopters operated in performance class 2 or 3 with MTOM of 3 175 kg or less, the FATO should be of sufficient size and shape to contain an area within which can be drawn a circle of diameter not less than 1 D. Note.— Local conditions, such as elevation and temperature, may need to be considered when determining the size of a FATO. 3.1.5 The mean slope in any direction on the FATO shall not exceed 3 per cent. No portion of a FATO shall have a local slope exceeding: a) 5 per cent where the heliport is intended to be used by helicopters operated in performance class 2 or 3 21. Annex 14 — Aerodromes Volume II 19/11/09 3-2 3.1.6 The surface of the FATO shall: a) be resistant to the effects of rotor downwash; b) be free of irregularities that would adversely affect the take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient to accommodate a rejected take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) have bearing strength sufficient take-off or landing of helicopters; and c) 1. 3.1.7 The surface of a FATO surrounding a touchdown and lift-off area (TLOF) intended for use by helicopters operated in performance classes 2 and 3 shall be static load-bearing. 3.1.8 Recommendation.— The FATO should provide ground effect. Helicopter clearways 3.1.9 When a helicopter clearway is provided, it shall be located beyond the end of the rejected take-off area available. 3.1.10 Recommendation.— The width of a helicopter clearway should not be less than that of the associated safety area. 3.1.11 Recommendation.— The ground in a helicopter clearway should not be less than that of the associated safety area. is located on the periphery of the FATO. 3.1.12 Recommendation.— An object situated on a helicopter clearway which may endanger helicopters in the air should be removed. Touchdown and lift-off areas 3.1.13 At least one TLOF shall be provided at a heliport. Note 1.— The TLOF may or may not be located within the FATO. Note 2.— Additional TLOFs may be collocated with helicopter stands. 3.1.14 The TLOF shall be of sufficient size to contain a circle of diameter of at least 0.83 D of the largest helicopter stands. 3.1.14 The TLOF shall be sufficient size to contain a circle of diameter of at least 0.83 D of the largest helicopter stands. surface of the area, but shall not exceed 2 per cent in any direction. 3.1.16 Where the TLOF is within the FATO, the TLOF shall be dynamic load-bearing and be capable of withstanding the traffic of helicopters that the area is intended to serve. 3.1.18 Where the TLOF is within the FATO, the centre of the TLOF shall be located not less than 0.5 D from the edge of the FATO. Safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 A FATO shall be surrounded by a safety areas 3.1.19 in performance class 1 in visual meteorological conditions (VMC) shall extend outwards from the periphery of the FATO is intended to serve and: a) each external side of the safety area shall be at least 2 D where the FATO is quadrilateral; or b) the outer diameter of the safety area shall be at least 2 D where the FATO is circular. 3.1.21 A safety area surrounding a FATO intended to be used by helicopters operated in performance class 2 or 3 in visual meteorological conditions (VMC) shall extend outwards from the periphery of the FATO for a distance of at least 3 m or 0.5 D, whichever is greater, of the largest helicopter the FATO is intended to serve and: a) each external side of the safety area shall be at least 2 D where the FATO is circular. 3.1.22 There shall be a protected side slope rising at 45 degrees from the edge of the safety area to a distance of 10 m, whose surface shall not be penetrated by obstacles, except that when obstacles are located to one side of the FATO only, they may be permitted to penetrate the side slope surface. 3.1.23 A safety area surrounding a FATO intended to be used by helicopter operations in instrument meteorological conditions (IMC) shall extend: a) laterally to a distance of at least 45 m on each side of the centre line; and b) longitudinally to a distance of at least 60 m beyond the ends of the FATO. (See Figure 3-1.) 3.1.24 No fixed object shall be permitted on a safety area, except for frangible objects, which, because of their function, must be located on the area. No mobile object shall be permitted on a safety area during helicopter operations. 3.1.25 Objects whose functions require them to be located along the edge of the FATO nor penetrate a plane originating at a height of 25 cm when located along the edge of the FATO and sloping upwards from the edge of the FATO at a gradient of 5 per cent. 3.1.26 Recommendation.— In the case of a FATO of diameter less than 1 D, the maximum height of 5 cm. Figure 3-1. Safety area should not exceed a height of 5 cm. Figure 3-1. Safety area should not exceed a height of 5 cm. m 60 m 90 m 23. Annex 14 — Aerodromes Volume II 19/11/09 3-4 3.1.27 The surface of the safety area, when solid, shall not exceed an upward slope of 4 per cent outwards from the edge of the FATO. 3.1.28 Where applicable, the surface of the safety area shall be treated to prevent flying debris caused by rotor downwash. 3.1.29 The surface of the the FATO shall be continuous with the FATO. Helicopter ground taxiways and ground taxiways and ground taxiway is intended to permit the surface movement of a wheeled helicopter ground taxiway is intended for the safety of simultaneous operations during the manoeuvring of helicopters. However, the wind velocity induced by the rotor downwash might have to be considered. Note 3.— When a taxiways for aeroplanes and helicopters, the provisions for taxiways for aeroplanes and helicopters. applied. 3.1.30 The width of a helicopter ground taxiway shall not be less than 1.5 times the largest width of the undercarriage (UCW) of the helicopter ground taxiway shall not exceed 3 per cent. 3.1.32 A helicopter ground taxiway shall be static load-bearing and be capable of withstanding the traffic of the helicopters the helicopter ground taxiway is intended to serve. Figure 3-2. Ground taxiway = 1.5 UCW Protection zone 24. Chapter 3 Annex 14 — Aerodromes 3-5 19/11/09 3.1.33 A helicopter ground taxiway shall be centred on a ground taxi-route. 3.1.34 A helicopter ground taxi-route shall extend symmetrically on each side of the centre line for at least 0.75 times the largest overall width of the helicopter ground taxi-route, except for frangible objects, which, because of their function, must be located there. 3.1.36 The helicopter ground taxiway and the ground taxi-route shall provide rapid drainage but the helicopter ground taxi-route shall be resistant to the effect of rotor downwash. Helicopter air taxi-route shall provide rapid drainage but the helicopter ground taxi-route shall be resistant to the effect of rotor downwash. helicopter air taxiway is intended to permit the movement of a helicopter above the surface at a height normally associated with ground effect and at ground speed less than 37km/h (20 kt). 3.1.38 The width of a helicopter air taxiway is intended to serve (see Figure 3-3). 3.1.39 The surface of a helicopter air taxiway shall be suitable for an emergency landing. 3.1.40 Recommendation.— The surface of a helicopter air taxiway shall be suitable for an emergency landing. 3.1.40 Recommendation.— The surface of a helicopter air taxiway shall be suitable for an emergency landing. 3.1.40 Recommendation.— The surface of a helicopter air taxiway shall be suitable for an emergency landing. Aerodromes Volume II 19/11/09 3-6 3.1.41 Recommendation.— The transverse slope of the surface of a helicopter air taxiway should not exceed 7 per cent. In any event, the slopes should not exceed 10 per cent and the longitudinal slope should not exceed 7 per cent. helicopter air taxiway shall be centred on an air taxi-route. 3.1.43 A helicopter air taxi-route shall extend symmetrically on each side of the centre line for a distance at least equal to the helicopters it is intended to serve. 3.1.44 No objects shall be permitted on an air taxi-route, except for frangible objects, which, because of their function, must be located thereon. 3.1.45 The surface of an air taxi-route shall be resistant to the effect of rotor downwash. 3.1.46 The surface of an air taxi-route shall provide ground effect. Air transit route is intended to permit the movement of a helicopter above the surface, normally at heights not above 30 m (100 ft) above ground level and at ground speeds exceeding 37 km/h (20 kt). 3.1.47 The width of an air transit route is intended to serve when the air transit route is intended for use by day only; and b) 10.0 times the largest overall width of the helicopters the air transit route is intended to serve when the air transit route is intended for use at night. 3.1.48 Any variation in the direction of the centre line of an air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route shall not exceed 120 degrees and be designed so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route so as not to necessitate a turn of radius less than 270 m. Note.— It is intended that air transit route so a autorotative or one-engine-inoperative landings such that, as a minimum requirement, injury to persons on the ground or water, or damage to property are minimized. Aprons 3.1.49 The slope in any direction on a helicopter stand shall not exceed 2 per cent. 3.1.50 A helicopter stand shall be of sufficient size to contain a circle of diameter of at least 1.2 D of the largest helicopter the stand is intended to serve. 3.1.51 If a helicopter stand is used for taxi-through, the minimum width of the stand and protection area shall be not less than 2 D (see Figure 3-5). 3.1.53 When a helicopter stand is used for turning, it shall be surrounded by a protection area which extends for a distance of 0.4 D from the edge of the helicopter stand. 26. Chapter 3 Annex 14 — Aerodromes 3-7 19/11/09 Figure 3-4. Helicopter stand is used for turning, it shall be surrounded by a protection area which extends for a distance of 0.4 D from the edge of the helicopter stand. Protection zone Central zone 27. Annex 14 — Aerodromes Volume II 19/11/09 3-8 3.1.54 For simultaneous operations, the protection area of helicopter stands and their associated taxiroutes may overlap (see Figure 3-7). 3.1.55 When intended to be used for ground taxi operations by wheeled helicopters the stand is intended to serve. 3.1.56 A helicopter stand and associated protection area intended to be used for air taxiing shall provide ground effect. 3.1.57 No fixed objects shall be permitted on a helicopter stand and the associated protection area. 3.1.58 The central zone of the stand shall be capable of withstanding the traffic of helicopter it is intended to serve; or b) for a helicopter stand intended to be used for ground taxi-through, the same width as the ground taxiway. Note.— For a helicopter stands designed for hover turns with air taxiroutes/taxiways — simultaneous operations 2 D 2 × largest overall width 28. Chapter 3 Annex 14 — Aerodromes 3-9 19/11/09 Figure 3-7. Helicopter stands designed for hover turns with air taxi-routes/taxiways — non-simultaneous operations Location of a final approach and take-off area in relation to a runway or taxiway 3.1.59 Where a FATO is located near a runway or taxiway, and simultaneous VMC operations are planned, the separation distance between the edge of a runway or taxiway and the edge of a runway or taxiway intersections or holding points where jet engine efflux is likely to cause high turbulence; or b) near areas where aeroplane vortex wake generation is likely to exist. 3.2 Elevated heliports Note 1.— The dimensions of the taxi-routes and helicopter stands include a protection area. Note 2.— Guidance on structural design for elevated heliports is given in the Heliport Manual (Doc 9261). 3.2.1 In the case of elevated heliports, design considerations of the heliport shall take into account additional loading resulting from the presence of personnel, snow, freight, refuelling, fire fighting equipment, etc. 2 D 2 × largest overall width 29. Annex 14 — Aerodromes Volume II 19/11/09 3-10 Final approach and take-off areas and touchdown and lift-off areas Note. — On elevated heliports it is presumed that the FATO shall be coincidental. 3.2.2 An elevated heliport shall be obstacle free. 3.2.4 The dimensions of the FATO shall be: a) where intended to be used by helicopters operated in performance class 1, as prescribed in the helicopter flight manual (HFM) except that, in the absence of width specifications, the width shall be not less than 1 D of the largest helicopters operated in performance class 2 or 3, of sufficient size and shape to contain an area within which can be drawn a circle of diameter not less than: 1) 1 D of the largest helicopters the FATO is intended to serve is 3 175 kg or less. 3.2.5 Recommendation.— Where intended to be used by helicopters operated in performance class 2 or 3 with MTOM of 3 175 kg or less, the FATO should be of sufficient size and shape to contain an area within which can be drawn a circle of diameter not less than 1 D. Note.— Local conditions, such as elevation and temperature, may need to be considered when determining the size of a FATO. Guidance is given in the Heliport Manual (Doc 9261). 3.2.6 Slopes on a FATO at an elevated heliport shall be sufficient to prevent accumulation of water on the surface of the FATO shall be: a) resistant to the effects of rotor downwash; and b) free of irregularities that would adversely affect the take-off or landing of helicopters. 3.2.9 Recommendation.— The FATO should provide ground effect. Helicopter clearway is provided, it shall be located beyond the end of the rejected take-off area available. 3.2.11 Recommendation.— The width of a helicopter clearway is provided, it shall be located beyond the end of the rejected take-off area available. should not be less than that of the associated safety area. 3.2.12 Recommendation.— When solid, the surface of the helicopter clearway should not project above a plane having an upward slope of 3 per cent, the lower limit of this plane being a horizontal line which is located on the periphery of the FATO. 30. Chapter 3 Annex 14 — Aerodromes 3-11 19/11/09 Table 3-1. FATO minimum separation distance If aeroplane mass and/or helicopter mass are Distance between FATO edge and runway edge or taxiway edge o Recommendation.— An object situated on a helicopter clearway which may endanger helicopters in the air should be removed. Touchdown and lift-off areas 3.2.14 One TLOF shall be coincidental with the FATO. Note.— Additional TLOFs may be collocated with helicopter stands. 3.2.15 For a TLOF coincidental helicopter stand shall be sufficient to prevent accumulation of water on the surface of the area, but shall not exceed 2 per cent in any direction. 3.2.18 When the TLOF is collocated with a helicopter stand and intended to be used by ground taxiing helicopters only, the TLOF shall at least be static load-bearing and be capable of withstanding the traffic of the helicopters the area is intended to serve. 3.2.19 When the TLOF is collocated with a helicopter stand and intended to be used by air taxiing helicopters, the TLOF shall have a dynamic load-bearing area. Safety areas 3.2.20 The FATO intended to be used by air taxing helicopters, the TLOF is collocated with a helicopter stand and intended to be used by air taxing helicopters. to be used by helicopters operated in performance class 1 in visual meteorological conditions (VMC) shall extend outwards from the periphery of the FATO is intended to serve and: 31. Annex 14 — Aerodromes Volume II 19/11/09 3-12 a) each external side of the safety area shall be at least 2 D where the FATO is quadrilateral; or b) the outer diameter of the safety area shall be at least 2 D where the FATO is circular. 3.2.22 A safety area shall be at least 2 D where the FATO is circular. periphery of the FATO for a distance of at least 3 m or 0.5 D, whichever is the greater, of the largest helicopter the FATO is circular. 3.2.23 There shall be at least 2 D where the FATO is circular. 3.2.23 There shall be at least 2 D where the FATO is circular. be a protected side slope rising at 45 degrees from the edge of the safety area to a distance of 10 m, whose surface shall not be penetrated by obstacles, except that when obstacles are located to one side of the FATO only, they may be permitted on a safety area, except for frangible objects, which, because of their function, must be located on the area. No mobile object shall be permitted on a safety area shall not exceed a height of 25 cm when located along the edge of the FATO nor penetrate a plane originating at a height of 25 cm above the edge of the FATO and sloping upwards and outwards from the edge of the FATO at a gradient of 5 per cent. 3.2.26 Recommendation.— In the case of a FATO of diameter less than 1 D, the maximum height of 5 cm. 3.2.27 The surface of the safety area, when solid, shall not exceed an upward slope of 4 per cent outwards from the edge of the FATO. 3.2.28 Where applicable, the surface of the safety area abutting the FATO shall be continuous with the FATO. Helicopter ground taxiways and ground taxi-routes Note.— The following specifications are intended for the safety of simultaneous operations during the manoeuvring of helicopters. However, the wind velocity induced by the rotor downwash might have to be considered. 3.2.30 The width of a helicopter ground taxiway shall not be less than 2 times the largest width of the undercarriage (UCW) of the helicopters the ground taxiway shall not exceed 3 per cent. 3.2.31 The longitudinal slope of a helicopter ground taxiway shall not exceed 3 per cent. ground taxiway is intended to serve. 3.2.33 A helicopter ground taxi-route shall be centred on a ground taxi-route shall extend symmetrically on each side of the centre line to a distance not less than the largest overall width of the helicopters it is intended to serve. 3.2.34 A helicopter ground taxi-route shall extend symmetrically on each side of the centre line to a distance not less than the largest overall width of the helicopters it is intended to serve. 19/11/09 3.2.35 No objects shall be permitted on a helicopter ground taxi-route, except for frangible objects, which, because of their function, must be located there. 3.2.36 The helicopter ground taxi-route shall not exceed 2 per cent. 3.2.37 The surface of a helicopter ground taxi-route shall be resistant to the effect of rotor downwash. Helicopter air taxiway is intended to permit the movement of a helicopter air taxiway is intended to permit taxiway of a helicopter air taxiway shall be at least three times the largest width of the undercarriage (UCW) of the helicopters the air taxiway shall be dynamic load-bearing. 3.2.40 The transverse slope of the surface of a helicopter air taxiway shall not exceed 2 per cent and the longitudinal slope shall not exceed 7 per cent. In any event, the slopes shall not exceed the slope landing limitations of the helicopter air taxi-route. 3.2.42 A helicopter air taxi-route shall extend symmetrically on each side of the centre line to a distance not less than the largest overall width of the helicopters it is intended to serve. 3.2.43 No objects shall be permitted on an air taxi-route shall be resistant to the effect of rotor downwash. 3.2.45 The surface of an air taxi-route shall provide ground effect. Aprons 3.2.46 The slope in any direction on a helicopter stand shall not exceed 2 per cent. 3.2.47 A helicopter stand is intended to serve. 3.2.48 If a helicopter stand is used for taxi-through, the minimum width of the stand and

associated protection area shall be that of the taxi-route. 3.2.49 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be surrounded by a protection area shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning, it shall be not less than 2 D. 3.2.50 When a helicopter stand is used for turning the standard be not less than 2 D. 3.2.50 When a helicopter standard be not less than 2 D. 3.2.50 When a helicopter standard be not less than 2 D. 3.2.50 When a helicopter standard be not less than 2 D. 3.2.50 When a helicopter standard be not less than 2 D. 3.2.50 When a helicopter standard be not less than 2 D. 3.2.50 When a helicopter standard be not less than 2 D. 3.2.50 When a helicopter standard be not less than 2 D. 3.2.50 When a helicopter standard be not less than 3 D. 3.2.5 helicopter stand. 33. Annex 14 — Aerodromes Volume II 19/11/09 3-14 3.2.51 For simultaneous operations, the protection area of helicopter stands and their associated taxi-routes shall not overlap. 3.2.52 For simultaneous operations, the protection area of helicopter stands and their associated taxi-routes shall not overlap. When intended to be used for ground taxi operations by wheeled helicopters, the dimensions of a helicopter stand shall take into account the minimum turn radius of the wheeled helicopters, the dimensions of a helicopter stand shall take into account the minimum turn radius of the wheeled helicopter stand shall take into account the minimum turn radius of the wheeled helicopters the stand is intended to serve. 3.2.53 A helicopter stand and associated protection area intended to be used for air taxiing shall provide ground effect. fixed objects shall be permitted on a helicopter stand and the associated protection area. 3.2.55 The central zone of the helicopter stand shall be capable of withstanding the traffic of the helicopter it is intended to serve; or b) for a helicopter stand intended to be used for ground taxi-through, the same width as the ground taxiing only shall be dynamic load-bearing. 3.2.57 The central zone of a helicopter stand intended to be used for ground taxi-through, the same width as the ground taxi-through. helicopter stand intended to be used for turning on the ground, the dimension of the central zone might have to be increased. 3.3 Helidecks Note.— The following specifications are for helidecks located on structures engaged in such activities as mineral exploitation, research or construction. See 3.4 for shipboard heliport provisions. Final approach and take-off areas and touchdown and lift-off areas Note.— On helidecks it is presumed that the FATO and the TLOF. Guidance on the effects of airflow direction and turbulence, prevailing wind velocity and high temperatures from gas turbine exhausts or flare radiated heat on the location of the FATO is given in the Heliport Manual (Doc 9261). 3.3.1 The specifications in 3.3.9 and 3.3.10 shall be applicable for helideck shall be of sufficient size to contain: a) for helicopters with an MTOM of more than 3 175 kg, an area within which can be accommodated a circle of diameter not less than 0.83 D of the largest helicopter the helideck is intended to serve. 34. Chapter 3 Annex 14 — Aerodromes 3-15 19/11/09 3.3.4 Recommendation.— For helicopters with an MTOM of 3 175 kg or less, the FATO should be of sufficient size to contain an area within which can be accommodated a circle of diameter not less than 1.0 D of the largest helicopter the helideck is intended to serve. 3.3.5 A FATO shall be dynamic load-bearing. 3.3.6 A FATO shall provide ground effect. 3.3.7 No fixed objects, which, because of their function, must be located thereon. 3.3.8 Objects whose function require them to be located on the edge of the FATO shall not exceed a height of 25 cm, except that in the case of a FATO of diameter less than 1 D, the maximum height of 5 cm. 3.3.9 Objects whose function requires them to be located within the FATO (such as lighting or nets) shall not exceed a height of 2.5 cm. Such objects may be present only if they do not represent a hazard to helicopters. Note.— Examples of potential hazards include nets or raised fittings on the deck that might induce dynamic rollover for helicopters equipped with skids. 3.3.10 Safety net or safety shelves shall be located around the edge of a helideck but shall not exceed the helideck height. 3.3.11 The surface of the FATO shall be skid-resistant to both helicopters and persons and be sloped to prevent pooling of water. Note.— Guidance on rendering the surface of the FATO skid-resistant is contained in the Heliports 3.4.1 The specifications in 3.4.11 shall be applicable to shipboard heliports and persons and be sloped to prevent pooling of water. Note.— Guidance on rendering the surface of the FATO skid-resistant is contained in the Heliport Manual (Doc 9261). 3.4 Shipboard heliports 3.4.1 The specifications in 3.4.11 shall be applicable to shipboard heliports and persons and be sloped to prevent pooling of water. Note.— Guidance on rendering the surface of the FATO skid-resistant is contained in the Heliport Manual (Doc 9261). 2012. 3.4.2 When helicopter operating areas are provided in the bow or stern of a ship or are purpose-built above the ship's structure, they shall be regarded as purpose-built shipboard heliports. Final approach and take-off areas and touchdown and lift-off areas and touchdown and lift-off areas and touchdown and lift-off areas are provided in the FATO and the TLOF will be coincidental. Reference to FATO within the shipboard heliport section of this Annex is assumed to include the TLOF. Guidance on the effects of airflow direction and turbulence, prevailing wind velocity and high temperature from gas turbine exhausts or flare radiated heat on the location of the FATO is given in the Heliport Manual (Doc 9261). 3.4.3 Shipboard heliports shall be provided with at least one FATO. 3.4.4 The FATO of a shipboard heliport shall be of sufficient size to contain a circle with a diameter not less than 1.0 D of the largest helicopter the heliport is intended to serve. 35. Annex 14 — Aerodromes Volume II 19/11/09 3-16 3.4.7 For purpose-built shipboard heliports provided in the bow or stern of a ship, the FATO shall be of sufficient size to: a) contain a circle with a diameter not less than 1 D of the largest helicopter the heliport is intended to serve; or b) for operations with limited touchdown directions, contain an area within which can be accommodated two opposing arcs of a circle with a diameter not less than 1 D in the helicopter's longitudinal direction. Figure 3-8. Shipboard Note 2.— The touchdown heading of the helicopter is limited to the angular distance subtended by the 1 D arc headings, minus the angular distance which corresponds to 15 degrees at each end of the arc. 3.4.8 For non-purpose-built shipboard heliports, the FATO shall be of sufficient size to contain a circle with a diameter not less than 1 D of the largest helicopter the helideck is intended to serve. 3.4.9 No fixed objects whose function, must be located thereon. 3.4.10 Objects whose function require them to be located on the edge of the FATO shall not exceed a height of 25 cm. 3.4.11 Objects whose function requires them to be located within the FATO (such as lighting or nets) shall not exceed a height of 2.5 cm. Such objects may be present only if they do not represent a hazard to helicopters. 3.4.12 The surface of the FATO shall be skid-resistant to both helicopters and persons. 37. ANNEX 14 — VOLUME II 4-1 19/11/09 CHAPTER 4. OBSTACLE RESTRICTION AND REMOVAL Note.— The objectives of the specifications in this chapter are to define the airspace around heliports to be conducted safely and to prevent the heliports becoming unusable by the growth of obstacles around them. This is achieved by establishing a series of obstacle limitation surfaces and sectors Approach surfaces and centred on a line passing through the centre of the FATO (see Figure 4-1). 4.1.2 Characteristics. The limits of an approach surface shall comprise: a) an inner edge horizontal and equal in length to the minimum specified width of the FATO plus the safety area, perpendicular to the centre line of the approach surface and located at the outer edge of the safety area; b) two side edges originating at the ends of the inner edge and: 1) for other than a precision approach FATO, diverging uniformly at a specified rate from the vertical plane containing the centre line of the FATO, to a specified height above FATO, and then diverging uniformly at a specified final width and continuing thereafter at that width for the remaining length of the approach surface; and c) an outer edge horizontal and perpendicular to the centre line of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the remaining length of the approach surface and at a specified final width and continuing thereafter at that width for the approach surface and at a specified final width and continuing thereafter at that width for the approach surface and at a specified final width and continuing thereafter at the approach surface and the approach surface at th FATO. 4.1.3 The elevation of the inner edge shall be the elevation of the safety area at the point on the inner edge that is intersected by the centre line of the surface. A.1.4 The slope(s) of the approach surface shall be measured in the vertical plane containing the centre line of the safety area at the point on the inner edge that is intersected by the centre line of the surface. in performance class 2 and 3, it is intended that approach paths be selected so as to permit safe forced landing or one-engine-inoperative landing or one-engine-inoperative landing such that, as a minimum requirement, injury to the occupants of the helicopter. The most critical helicopter type for which the heliport is intended and the ambient conditions will be factors in determining the suitability of such areas. Transitional surface 4.1.5 Description. A complex surface along the side of the safety area and part of the side of the approach surface, that slopes upwards and outwards to the inner horizontal surface or a predetermined height (see Figure 4-1). 38. Annex 14 — Aerodromes Volume II 19/11/09 4-2 4.1.6 Characteristics. The limits of a transitional surface with the inner horizontal surface, or beginning at a specified height above the lower edge when an inner horizontal surface is not provided, and extending down the side of the approach surface to the inner edge of the safety area parallel to the centre line of the safety area parall specified height above the lower edge when an inner horizontal surface is not provided. 4.1.7 The elevation of the approach surface at that point; and b) along the safety area — equal to the elevation of the elevation of the elevation of the approach surface at that point; and b) along the safety area — equal to the elevation of the elevation of the approach surface at that point area. point. Note.— As a result of b) the transitional surface along the safety area will be curved if the profile of the FATO is curved, or a plane if the profile of the transitional surface with the inner horizontal surface. depending on the profile of the FATO. 4.1.8 The slope of the transitional surface is to allow safe visual manoeuvring. 4.1.9 Description. A circular surface located in a horizontal plane above a FATO and its environs (see Figure 4-1). 4.1.10 Characteristics. The radius of the inner horizontal surface shall be measured above an elevation datum established for such purpose. Note.— Guidance on determining the elevation datum is contained in the Heliport Manual (Doc 9261). Conical surface 4.1.12 Description. A surface sloping upwards and outwards from the periphery of the inner horizontal surface is not provided (see Figure 4-1). 4.1.13 Characteristics. The limits of the conical surface shall comprise: a) a lower edge coincident with the periphery of the inner horizontal surface, or outer limit of the transitional surface is not provided; and b) an upper edge located at a specified height above the inner horizontal surface is not provided. 39. Chapter 4 Annex 14 — Aerodromes 4-3 19/11/09 4.1.14 The slope of the conical surface shall be measured above the horizontal. Take-off climb surface 4.1.15 Description. An inclined plane, a combination of planes or, when a turn is involved, a complex surface 4.1.15 Description. centre of the FATO (see Figure 4-1). 4.1.16 Characteristics. The limits of a take-off climb surface shall comprise: a) an inner edge horizontal and equal in length to the minimum specified width of the FATO plus the safety area or clearway; b) two side edges originating at the ends of the inner edge and diverging uniformly at a specified rate from the vertical plane containing the centre line of the take-off climb surface and at a specified height above the elevation of the FATO. 4.1.17 The elevation of the inner edge shall be the elevation of the safety area at the point on the inner edge that is intersected by the centre line of the take-off climb surface, except that when a clearway is provided, the elevation shall be equal to the highest point on the ground on the centre line of the clearway. be measured in the vertical plane containing the centre line of the surface. 4.1.19 In the case of a take-off climb surface involving a turn, the surface shall be the same as that for a straight take-off climb surface. That portion of the surface between the inner edge and 30 m above the inner edge shall be straight. 4.1.20 Any variation in the direction of the centre line of a take-off climb surface shall be designed so as not to necessitate a turn of radius less than 270 m. Note.— For heliports used by helicopters operated in performance class 2 and 3, it is intended that departure paths be selected so as to permit safe forced landings or one-engine-inoperative landings such that, as a minimum requirement, injury to persons on the ground or water or damage to property are minimized. Provisions for forced landing areas are expected to minimize risk of injury to the occupants of the helicopter. The most critical helicopter type for which the heliport is intended and the ambient conditions will be factors in determining the suitability of such areas. Obstacle-free sector/surface — helidecks 4.1.21 Description. A complex surface originating at and extending from a reference point shall be located not less than 0.5 D from the centre of the FATO. 4.1.22 Characteristics. An obstacle-free sector/surface shall subtend an arc of specified angle. 4.1.23 A helideck level (see Figure 4-2): 40. Annex 14 — Aerodromes Volume II 19/11/09 4-4 a) Above helideck level. The surface shall be a horizontal plane level with the elevation of the helideck surface that subtends an arc of at least 210 degrees with the apex located on the periphery of the D reference circle extending outwards to a distance that will allow for an unobstructed departure path appropriate to the helideck is intended to serve. b) Below helideck level. Within the (minimum) 210-degree arc, the surface shall additionally extend downward from the edge of the FATO below the elevation of the helideck to water level for an arc of not less than 180 degrees that passes through the centre of the FATO and outwards to a distance that will allow for safe clearance from the obstacles below the helideck in the event of an engine failure for the type of helicopter the helideck is intended to serve. Note.— For both the above obstacle-free sectors for helicopters operated in performance class 1 or 2, the helicopters operated in performance class 1 or 2, the helicopter sectors for helicopter the helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopter sectors for helicopters operated in performance class 1 or 2, the helicopters operated in performance class 1 or 2, the helicopters operated in performance class 1 or 2, the helicopters operated in perform type to be used. Limited obstacle sector/surface — helidecks Note.— Where obstacles are necessarily located on the structure, a helideck may have a limited obstacle sector. 4.1.24 Description. A complex surface originating at the reference point for the obstacle-free sector within which the height of obstacles above the level of the FATO will be prescribed. 4.1.25 Characteristics. A limited obstacle sector shall not subtend an arc greater than 150 degrees. Its dimensions and location surfaces are specified on the basis of the intended use of a FATO, i.e. approach manoeuvre to hover or landing, or take-off manoeuvre and type of approach, and are intended to be applied when such use is made of the FATO. In cases where operations are conducted to or from both directions of a FATO, then the function of certain surfaces may be nullified because of more stringent requirements of another lower surface. Surface-level heliports 4.2.1 The following obstacle limitation surface; b) approach surface shall be established for a nonprecision approach FATO: a) take-off climb surface; 41. Chapter 4 Annex 14 — Aerodromes 4-5 19/11/09 c) transitional surface; and d) conical surface; and d) conical surface; and d) conical surface; 41. Chapter 4 Annex 14 — Aerodromes 4-5 19/11/09 c) transitional surface; and b) approach surface. 4.2.4 Recommendation.— The following obstacle limitation surfaces should be established for a non-precision approach FATO: a) inner horizontal surface may not be required if a straight-in non-precision approach is provided at both ends. 4.2.5 The slopes of the surfaces shall not be greater than, and their other dimensions not less than those specified in Tables 4-1 to 4-4 and shall be located as shown in Figures 4-4 to 4-8. 4.2.6 New objects or extensions of existing objects shall not be permitted above any of the surfaces in 4.2.1 to 4.2.4 except when, in the opinion of the appropriate authority, the new object or extension would be shielded by an existing immovable object. Note.— Circumstances in which the shielding principle may reasonably be applied are described in the Airport Services Manual, Part 6 (Doc 9137). 4.2.7 Recommendation.— Existing objects above any of the surfaces in 4.2.1 to 4.2.4 should, as far as practicable, be removed except when, in the opinion of the appropriate authority, the object is shielded by an existing immovable object, or after aeronautical study it is determined that the object would not adversely affect the regularity of operations of helicopters. Note.— The application of curved take-off climb surfaces as specified in 4.1.19 may alleviate the problems created by objects infringing these surfaces. 4.2.8 A surface-level heliport is not less than 150 degrees. 4.2.9 Recommendation.— The number and orientation of take-off climb and approach surfaces should be such that the usability factor of a heliport is not less than 95 per cent for the helicopters the heliports shall conform to the requirements for surface-level heliports shall conform to the requirements for surfaces separated by not less than 150 degrees. 42. Annex 14 — Aerodromes Volume II 19/11/09 4-6 Helidecks Note.— The following specifications are for helideck may have a structure and engaged in such activities as mineral exploitation, research, or construction, but excluding heliports on ships. 4.2.12 A helideck shall have an obstacle-free sector. Note.— A helideck may have a limited obstacle sector (see 4.1.25). 4.2.13 There shall be no fixed obstacle-free sector above the obstacle-free surface. 4.2.14 In the immediate vicinity of the helideck, obstacle protection for helicopters shall be provided below the helideck. centre of the FATO, with a descending gradient having a ratio of one unit horizontally to five units vertically from the edges of the FATO within the 180-degree sector. This descending gradient may be reduced to a ratio of one unit horizontally to three within the 180-degree sector. Figure 4-2). 4.2.15 Where a mobile obstacle or combination of obstacle surface/sector out to a distance of 0.62 D, measured from the centre of the FATO, objects shall not exceed a height of 0.05 D above the FATO. Beyond that arc, out to an overall distance of 0.83 D the limited obstacle surface rises at a rate of one unit vertically for each two units horizontally (see Figure 4-3). Shipboard heliports located forward or aft 4.2.17 The specifications in 4.2.20 and 4.2.22 shall be applicable for shipboard heliports completed on or after 1 January 2012. 4.2.18 When helicopter operating areas are provided in the bow or stern of a ship, they shall apply the obstacle criteria given in 4.2.12, 4.2.14 and 4.2.16. Amidships location 4.2.19 Forward and aft of the FATO shall be two symmetrically located sectors, each covering an arc of 150 degrees, with their apexes on the periphery of the FATO D reference circle. Within the area enclosed by these two sectors, there shall be no objects rising above the level of the FATO. cm. 4.2.20 Objects whose function requires them to be located within the FATO (such as lighting or nets) shall not exceed a height of 2.5 cm. Such objects may be present only if they do not represent a hazard to helicopters. Note.— Examples of potential hazards include nets or raised fittings on the deck that might induce dynamic rollover for helicopters equipped with skids. 4.2.21 To provide further protection from obstacles fore and aft of the FATO, rising surfaces with gradients of one unit vertically to five units horizontally shall extend from the entire length of the edges of the two 150-degree sectors. These surfaces shall extend for a horizontal distance equal to at least 1 D of the largest helicopter the FATO is intended to serve and shall not be penetrated by any obstacle (see Figure 4-9)

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